

NOMANCLAURE LIST

10	WOVEN PI PREFORM
12&14	UPSTANDING LEGS
16&17	BOTTOM FOOT PORITONS
18	ROOT PORTION
20	WARP FIBERS
21	FIBERS PERPENDICULAR TO THE STANDING LEGS
22A&22B	ARROWS POINT TO CURVATURE
24	ARROW INDICATING BEND DIRECTION
26	CRUVED LENGTH OF PREFORM 10A
28	RADIUS OF CURVE
30	DIE
31	CUTTER HEAD
32	BLADES
33	WIDTH OF BLADES
35	SPACING OF CUTS
40	DIE ASSEMBLY
41	DIE HALF
42	DIE HALF
43	FORMING SURFACE
44	FORMING SURFACE
47	TAPERED END FORMING SURFACE 43
48	TAPERED END FORMING SURFACE 44
52A&52B	STRUCTURAL ELEMENTS
10B	PREFORM
41A	DIE HALF
42B	DIE HALF
43A	FORMING SURFACE OF DIE HALF 41A
44B	FORMING SURFACE OF DIE HALF 42A
10C	PREFORM
41B	DIE HALF
42B	DIE HALF
43B	FORMING SURFACE OF DIE HALF 41B
44B	FORMING SURFACE OF DIE HALF 42B

PROCESS FOR MAKING A CURVED PI SHAPED PREFORM MADE FROM WOVEN COMPOSITE MATERIALS

BACKGROUND OF THE INVENTION

Field of the Invention

The invention relates to the field of composite structure fabrication techniques and, in particular, to a process for forming curves in woven composite preforms.

Description of Related Art

Typically T shaped composite structures are fabricated by joining the vertical member to the horizontal member by bonding a multi-number of reinforcing sheets across the joint (extending from the horizontal member up along the side of the vertical member). Such a joint is disclosed in WIPO Publication WO 01/64387 A1 Production, Forming, Bonding Joining And Repair Systems For Composite And Metal Components by N. Graham. Two honeycomb sheets are joined by layers of composite cloth to the horizontal member on each side of the vertical member that extend up each side of the vertical member. The disadvantage is that the joint's strength is dependent on the layers of composite cloth.

Recently, three-dimensional weaving has allowed very complex shapes to be woven. For example, US Patent No.: 6,007,319 Continuous Forming Of Complex Molded Shapes by T. L. Jacobson, et al. discloses a method of weaving complex preform shapes. More recently a process for making woven 3D PI cross-section shapes in US Patent No. 6,446,675 Minimum Distortion 3D Woven Preforms by J. Goering. Such preforms can be impregnated with a resin and partially cured (called B stage) and stored for relatively long periods

1 at low temperature until use is required. However, attempts to use such a
2 preform in a curved structure has resulted in severe distortion.

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4 Thus, it is a primary object of the invention to provide a process for
5 making curved 3D woven PI preforms structures.

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7 It is a further object of the invention to provide a process for making
8 such 3D woven PI preforms in curved structures that does not significantly
9 reduce the strength of the preform.

10 11 **SUMMARY OF THE INVENTION**

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13 The invention is a process for forming a 3D woven PI shaped cross-
14 section preform having a first and second upstanding leg portions and first and
15 second foot portions for use in a structure having at least one curved portion of
16 a specific length. The process includes the steps of:

17 1. Cutting the threads parallel to the direction of curvature over a length equal
18 to the length of the curve, such that the cuts in each thread are spaced from
19 the cuts in the adjacent threads. Preferably, the first and second upstanding leg
20 portions are folded over the first and second bottom foot portions, prior to the
21 step of cutting.

22 2. Stretching the portions of the preform requiring curvature. If the preform
23 must be curved in the plane of the bottom foot portions with the first bottom
24 portion requires greater stretching than the second bottom foot portion, both
25 upstanding legs are bent over on to one of the bottom foot portions in a
26 "cactus" configuration. The preform is then placed between matched tapered
27 sign-wave dies, with a small amplitude end and a large amplitude end, with the
28 first portion positioned in the small amplitude end and the second end in the
29 large amplitude end. Stretching is accomplished by closing the die halves. If
30 the completed preform requires concave curvature of the bottom foot portions,
31 the bottom foot and upstanding leg portions of the preform are bent toward

each other in an "H" configuration, but only the upstanding legs are placed in a small to large amplitude tapered sign wave die for stretching while the foot portion is stretched at a constant amount in the large amplitude section of the die. If on the other hand the bottom foot portions require a convex shape, the bottom foot and upstanding leg portions are again folded together in an "H" configuration. However the die shape is tapered over the length of the upstanding legs from small to large amplitude and the foot bottom portions are outside of the stretching sinewave die.

3. Forming the curvature in the preform. After the step of stretching, the preform is expanded about a die surface having the final desired shape of the preform.

The preform can thereafter be used in the making of curved composite structures, primarily as a transition member between sheet type structural members. Note that the preform can be pre-impregnated with a resin prior to any forming steps.

The novel features which are believed to be characteristic of the invention, both as to its organization and method of operation, together with further objects and advantages thereof, will be better understood from the following description in connection with the accompanying drawings in which the presently preferred embodiment of the invention is illustrated by way of example. It is to be expressly understood, however, that the drawings are for purposes of illustration and description only and are not intended as a definition of the limits of the invention.

BRIEF DESCRIPTION OF THE DRAWINGS

Figure I is a perspective view of the PI shaped preform made of woven filamentary material.

1 Figure 2 is a top view of the PI shaped preform shown in Figure 1
2 illustrating the effect of attempting to curve the preform in the “as is” condition.

3
4 Figure 3 is a perspective view of the desired curved shape for the
5 preform shown in Figure 1.

6
7 Figure 4 is an end view of the preform with the legs folded over on the
8 bottom portions installed in a darting die assembly.

9
10 Figure 5 is a side view of the bottom portion of the die illustrating the
11 placement of the cutters used for darting the preform.

12
13 Figure 6 is an enlarged view of a portion of the preform, after darting
14 illustrating the darting pattern.

15
16 Figure 7 is a perspective view of a sine wave forming die assembly
17 used for selectively stretching the preform.

18
19 Figure 8 is a cross-sectional view of the die assembly shown in Figure 7
20 illustrating a first method of stretching in order to form the preform shown in
21 Figure 3.

22
23 Figure 9 is a cross-sectional view of a completed structure using the
24 preform shown in Figure 3.

25
26 Figure 10 is a perspective view of the preform having a convex
27 curvature.

28
29 Figure 11 is a cross-sectional view of the die assembly shown in Figure
30 7 illustrating a second method of stretching the preform to obtain the curvature
31 shown in Figure 10.

Figure 12 is a perspective view of the preform having a concave curvature.

Figure 13 is a cross-sectional view of the die assembly shown in Figure 7 illustrating a third method of stretching the preform to obtain the curvature shown in Figure 12.

DESCRIPTION OF THE PREFERRED EMBODIMENT

The typical PI woven preform is illustrated in Figures 1 and 2, indicated by numeral 10. The preform 10 includes upstanding legs 12 and 14 and bottom foot portions 16 and 17, with center or root 18. The warp fibers 20 run parallel to the legs 12 and 14, while the fibers 21 run perpendicular to the upstanding legs. If one tries to form a curve, indicated by arrows 22A and 22B, the legs 12 and 14 tend to bend over indicated by arrow 24. Attempts to "bend" the preform 10 into other shapes also cause one or more portions to warp. The subject process will eliminate this problem.

A completed preform 10A is shown in Figure 3, having a curved length 26, with a radius 28. To accomplish this, the upstanding legs 12 and 14 are pushed over onto the one of the bottom portions 16 or 17 as shown in Figure 14. The folded preform is then placed in the die 30 shown in Figures 4 which includes a cutter head 31 and a receiver pad 32. The cutter head 31 incorporates staggered blades 32 having a width 33 as shown in Figure 5 slightly greater than the width of the warp threads 20, allowing for some mismatch in warp thread location. This allows the warp threads 20 to be cut (dashed) periodically into segments such that the cuts in each tread are spaced from the cuts in the adjacent treads as shown in Figures 6. The spacing 35 of the cuts should be as large a distance as possible, but still allowing the

1 curved length 26 to be formed. Thus some experimentation may be required
2 to obtain the optimum spacing.

3
4 Referring to Figures , if the part is to be simply curved shape as shown
5 in Figure3, the darted preform 10 is folded as shown in Figure 14 Cactus with
6 the legs 12 and 14 bent over on to leg 16. The preform 10 is placed in a sine-
7 wave shaped die assembly 40 having matched die halves 41 and 42 with
8 mating sign-wave shaped forming surfaces 43 and 44 respectively. The sign-
9 wave pattern is on forming surface 43 is tapered from ends 45 and 46 on die
10 half 41 and the forming surface 44 is tapered from ends 47 and 48 on forming
11 surface 44. What the sine wave forming accomplishes is a stretching that is
12 zero at the end of bottom portion 17 and a maximum at the end of bottom
13 portion 16.

14
15 The now stretched preform 10 can be placed in a die assembly (not
16 shown) and formed into its final shape. Alternately the stretched preform can
17 be shaped by hand. Referring to Figure 9, it can thereafter be resin infused by
18 any of several existing resin infusion processes and be used to join to
19 structural elements together. For example, structural elements 52A and 52B,
20 by the process set forth in WIPO Publication WO 01/64387 A1 Production,
21 Forming, Bonding Joining And Repair Systems For Composite And Metal
22 Components by N. Graham. Of course, the preform could be resin infused
23 prior to darting and stretching.

24
25 If the completed preform requires curvature in a convex shape as
26 illustrated in Figure 10 and designated by numeral 10B, the preform 10 is
27 folded the shape as illustrated in Figure 11 with the legs 12 and 14 folded
28 together and portions 16 and 17 folded together. As illustrated the die halves
29 41A and 42A have forming 43A and 44A. Stretching would only from the
30 center outward toward the end of the legs 12 and 14 where stretching would
31 be at a maximum.

1
2 If on the other hand, the preform final shape shown in Figure 12, and
3 designated by numeral 10C, is desired, then, as illustrated in Figure 13, the
4 legs 12 and 14, and portions 16 and 17 are brought together as in the previous
5 example, and placed in the die assembly 40B having die halves 41B and 42B
6 with forming surfaces 43B and 44B. However, stretching is accomplished by
7 placing the folded preform 10 in the sine-wave dies such stretching of the legs
8 12 and 14 is a minimum at there ends and becomes a maximum at the center.
9 Thereafter, stretching of the bottom portions 16 and 17 is held constant.

10
11 Thus it can be seen that the process will allow the PI shaped preform to
12 molded into numerous curved shapes, many more than have been described
13 herein. While there is a weakening of the preform due to the cutting of the
14 warp fiber, the loss of strength has proven acceptable in most applications,
15 particularly where the primary loads and distributed along the fill fiber.

16
17 While the invention has been described with reference to a particular
18 embodiment, it should be understood that the embodiment is merely
19 illustrative, as there are numerous variations and modifications, which may be
20 made by those skilled in the art. Thus, the invention is to be construed as
21 being limited only by the spirit and scope of the appended claims.

22 23 **INDUSTRIAL APPLICABILITY**

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25 The invention has applicability to industries manufacturing composite
26 structures, particularly, the aircraft industry.